

TECHNICAL DEPT.

AVIATION

The Oldest American Aeronautical Magazine

APRIL 23, 1928

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XXIV

Special Features

NUMBER
17

Potomac Flying Service, Inc.
Airplane Dopes and Their Application
German Plane Crosses North Atlantic East to West

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Index to Contents

EDITORIAL	1145
FOREWORD BY THE EDITOR	1146
WING ANALYSIS	1146
AIRPLANE DOWN AND THEIR ASSURANCE	1146
CHERRY PLANE CROSSING NORTH ATLANTIC	1146
EAST TO WEST	1150
NEWS	1150-55
LAST MONTH EVENTS	1154
NOTES	1155
POWER NEWS	1159
ARMED AND AIRWAYS	1170
U S AIR FORCE	1175

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APR 1938



The Oldest American Aeronautical Magazine

Vol. XXIV

APRIL 23, 1938

No. 17

Jockey Pilots

PRIORITY PLANE designers devote countless time and effort to refinements in design which will save five pounds of weight and will cut off a few inches from the frontal area of the plane. Yet, while doing this they realize that they must design their plane to fit a man who is over six feet tall and who may weigh over two hundred pounds in his flying equipment and parachute. The answer seems obvious, cut down the size and weight of your pilot and you can do more than you can by any amount of refinements in design. For a certain kind of high speed pursuit plane the answer should undoubtedly mean recruiting and selecting pilots who have the same general characteristics as our top track jockeys.

The English have recently given a most interesting account of what could be done along this line. The Handley Company designed its Tiger Moth as is the cross section outline of their fuselage the smaller dimensions into which could be fitted a very small but excellent pilot. The company used a four cylinder in line air-cooled engine of only 130 hp. This is less than one-third of the power of most pursuit planes, yet both in high speed and in climb the plane equaled or exceeded the performance of any standard pursuit plane.

Our Army regulations now prescribe minimum height and weight requirements for enlistment. It may be as yet very distant date that the requirement for enlistment in the crash branch of our Air Corps will be a minimum instead of a maximum one. The development of a group of "Jockey Pilots" whose especially designed planes could outspeed those of any other part of the pursuit group would undoubtedly take a long time in coming but they would prove invaluable for reconnaissance in the enemy's territory and even though their movement were limited to one gun they would have a real offensive power.

Manufacturing Costs

ONE OF the interesting and curious features of the present day airplane manufacturing situation is that apparently the cost of the planes is not going down as rapidly as the increase in production. Extensive figures show that planes are as expensive as they were a few years ago yet the number built has increased tenfold. How do you know, say that this is not due to undue profits on the part of the manufacturers, but rather to enormous increase in the manufacture of airplanes as they are produced.

When airplanes built for commercial purposes first begin to compete with the war surplus stock they were

built in the back yard, so to speak. Only the simplest machinery was used, and a good carpenter and welder could turn out an excellently finished plane. The designer was also a test pilot, shop foreman and salesman, so there was practically no overhead. There were no dealers, so less had to be given out in commissions and many of the sales were direct. Due to the competition of the cheap War surplus planes the sale price had to be very low and the builder of the plane was having it made enough to keep himself going and paid himself a small salary. Had it not been for the War surplus competition the price would have started higher.

At present the man in his engineering and drafting, shop superintendent and stock room clerk, has to design and build jigs and to rent a tool factory. His dealers' commissions and his sales expenses have gone up greatly. In other words the direct labor cost may have gone down with production, but apparently the increase in overhead has just about counterbalanced this decrease. At what point increased production will allow of a real decrease in price to the consumer is hard to tell.

In many ways the situation is satisfactory as far as the development of aviation goes, for it means that a brilliant designer can bring out a new design, and get a start without an enormous financial backing. The manufacturers who have already reached a considerable production will, however, have an enormous advantage and so in other industries they will gradually force ahead the small producers will drop out of the picture.

With the gradual perfecting of metal planes which lead themselves to production methods this initial advantage will gradually be counteracted and once really large production is reached it will be as hard to break into airplane manufacturing as it is to break into automobile manufacturing at the present time.

Air Transport Advertising

IT IS to be admitted that the greatest selling point as regards air transportation is the item of speed, but at the same time air transport operators who are catering to passengers should bear in mind, when mulling out advertising campaigns, that there are other sales items of almost equal importance.

For the one who has set to take his first ride in an airplane there awaits an experience that cannot be equaled by any other means of transportation. The beauty and interest of the ever changing panorama below him is something that he will remember for a long time to come. It has often been stated that the appeal of all the beauties of Europe can be found within the borders of the United States. To view them from the best vantage point—the air—is a sight less than can be well said in air transport advertising.

Potomac Flying Service Inc.

Washington, D. C. Company Has Flown 15,000 Sight-Seeing Passengers Within Six Months of Its Organization

By AARON M. ROSENBLEET

FIFTEEN THOUSAND sight-seeing passengers within six months of its organization is the record of the Potomac Flying Service, Inc., Washington, D. C. Within that period the company's flying equipment increased from two to six planes including two four-passenger planes and the personnel increased from two pilots and one mechanic to five pilots, four mechanics, business managers, stenographers, secretary, bookkeeper, two collectors and a special ticket collector and passenger agent. This record was achieved not by hair-raising stunts and publicity but by the utilization of sound methods and principles which proved their worth and value in other commercial fields.

The Potomac Flying Service is located at Hoover Field, South Washington, Va., only twelve minutes ride from the heart of the National Capital. Henry A. Berlier of the Berlier Aircraft Co. is president of this company. The other officers of the concern are: Lewis H. Harding, vice president and general manager; Andrew H. Nash, secretary; and John D. Shook, treasurer. Mr. Nash also acts as publicity man and advertising manager. The company is engaged in sight-seeing and cross-country flying, visits places for photographic work and also manages a flying school with an average enrollment of 75 students.

No Spectacular Advertising

When Mr. Nash was asked by what means the company achieved the record, so far as statistics are available, of carrying the largest number of sight-seeing flyers by any aviation organization he gave credit to the policies set down by the organizers of the concern.

"The company has established itself by adherence to non-spectacular policies and sound business methods," he stated. And then he continued with his explanation.

"No spectacular advertising has been resorted to; it has never held an air show or stunts in an effort to attract passengers or publicity. Plans have never carried passengers except under favorable weather conditions and the passenger flights have been conducted without stop landing, clouds or glides. By pursuing this policy the company has had no accident and due to the careful maintenance of its planes and the selection of its mechanics and pilots this accident record should continue."

That statement gives only the general picture of the Potomac Flying Service. To find the specific principles whereby these policies were carried out and brought such satisfactory results, a careful analysis must be made of the methods used in attracting trade.

Washington is one of the leading sight-seeing and show places of the United States. Thousands of tourists either individually or under local agencies come weekly to Washington. The problem with the Potomac Flying Service, was what means to use to capture the tourist trade

A scientific analysis was made of the directory of a town in Washington. Every contact made by tourists in Washington was studied, and the following was discovered:

The visitors to Washington come in contact with the representatives of their tourist agency, hotel clerks, sight-seeing bus and taxi drivers, newspaper agents, insurance salesmen, and roadside store-keepers. It was also found that tourists hardly spend any time reading newspapers and when they do buy a paper it is only to see the front page and possibly the society page to glance at the "society dailies" of the fashionable circles. The tourist's attention is fastidious to read in newspaper and also the traffic expense of daily advertising in five newspapers eliminated that medium as a means to reach them with the message "to see Washington from the air."

Use of word advertising used effectively by Potomac Flying Service, Inc.

It was then decided to use more direct means of advertising the effect of the message and the attention of tourists in Washington.

Before the decision was carried out into effect the Potomac Flying Service had only a Saturday-day flying service with a average of 200 passengers for both days. During week day there was a declining interest which at times hardly paid in the morning up of the engine, the pilot's time and fuel.

As soon as the new policy of reaching the tourist was begun to be carried out the effect was almost immediate. At example taken at random from one Wednesday is quite shown the effect of the scientific analysis of the particular problem in Washington and which can be applied in all American city of any size.

Wednesday, 22; Thursday, 23; Friday, 24; Saturday, 25; Sunday, 26; Monday, 27; Tuesday, 28 (observed) and Wednesday, 29. This makes an average of 100 per day, a figure which would place many as any passenger where it is considered that a ride means five dollars per passenger.

April 23, 1934

The message to divert the steady tourist flow to Hoover Field was divided into three parts, direct printed advertising, word advertising and newspaper publicity.

First advertising. It was found that of the show places in Washington, two are never missed by a tourist. They are the Bureau of Engineering and the Arlington National Memorial. At either leaflet beautifully colored in sky blue with an orange bar showing the Washington Monument and the National Capital on the outside with a wrap-up on the inside was given up by a special commercial agent to immediately attract the eye, attract interest and create a desire to find out in leaflet's message.

"From the very inception of our advertising campaign we desired," Mr. Nash said, "to make every effort to send out



Exhibition of a flight leaflet. Note perforations on the left side.

leaflet type of advertising which will be attractive to the eye. Fly advertising matter only drives people away. Cheap looking material means a cheap concern. The impact and the work for the leaflet cost at \$250 and is one of the best mediums we have."

In the inside of the leaflet is a picture of the Arlington Memorial from the air and also an air photograph of the Capital. The message matter is a persuasive appeal to the tourist to see the location of Washington from the air. It is the record of the company, mentioning the problem as to use of advertising in its potential and safety in flying. Set off with a thin orange border is the picture of the "Flying Certificate" which forms a part of the flight leaflet and is left with the passenger as a memento of the air voyage over the city of Washington.

Tickets in Books of 50

The flight tickets are made up of clouded blue paper with a gold figure of an airplane and the wings which are the wings of the Potomac Flying Service. The tickets are in books of 50 and are packed securely. These tickets are distributed among the tourist agency offices, hotel clerks, newspaper agents—all responsible persons. When a ticket is sold the date is left in the book, which then serves as a guide to indicate the number of tickets sold by that particular agent. It is the part of the ticket is kept by the pilot which is held in the end of the day and serves as a guide for key and remittance. The number of perforations that the ticket makes on his part of the ticket indicates the number of passengers who went up on a trip. The third part is the "Flight Certificate" which states that on such a day made on a voyage over the city of Washington. The printed signature of the general manager is at the bottom on one side with the printed words, Hoover Field, South Washington, Va., and on the bottom on the other side. On top of the certificate, the spread wings of the Potomac Flying Service are shown. The wings are made of the middle, all of which have a very artistic effect on the clouded blue paper and a message which is actually worthwhile keeping.

Mr. Nash mentioned that visitors often come from all over the United States with the "Flight Certificate" these flights begin home with them and seldom to be taken up so they

AVIATION

1147

too could show a flight certificate like that when they return. The picture of the flight certificate on the leaflet brings many tourists to the flying field, and to make certain that they will not lose their way a diagram is drawn on one side of the leaflet to show the visitors how they can reach the field by either bus or trolley. On the same diagram a circle is drawn in gold showing the points of interest the plane flies over when making a trip, which includes the Lincoln Memorial, Washington Monument, White House, Museum, the heart of Washington, Potomac Park and Arlington.

Every day two collectors are sent out with a thousand leaflets each to distribute them at the Bureau of Engineering and Arlington. The leaflets are also distributed among the tourist office, tourist selling agencies, hotel clerks, news stands, restaurants and sight-seeing buses. The message certifies that the leaflet will be distributed a "society" card was printed stating that "The Potomac Flying Service Inc. will extend every courtesy to the friends of Mr. Nash." The name of the person who gives out the courtesy card is written in that space. This serves a double purpose. First it demonstrates the previous distribution of the ticket books which if lost would create a loss of a hundred and fifty dollars and at the same time creates a desire in the bus and taxi drivers to distribute them. The Potomac Flying Service gives the same commission to the person whose name is on the courtesy card as it does to the agency which sells a regular flight ticket. It was also found that many visitors who fly sight of the leaflet, accept the courtesy cards and go out to the field with the feeling that with that card they will receive special attention. These cards are kept, and at the end of each week the person whose name is on the cards are paid their commissions. These courtesy cards are left wherever the



The center spread of the school circular used by Potomac Flying Service, Inc.

leaflets are placed for distribution and are also given in all taxi drivers who come out to the field or who are reached by the "tourist man."

A special show made of two men was drawn up and it is an artistic blending of purple, orange, yellow and green which would catch the eye from the distance. A mass design at the top made for the purpose of attracting the curiosity. These cards have very little reading matter. "Fly over Washington" they suggest and give the name of the company, its location and price for a flight. These cards were placed in all the offices handling the company's tickets, hotel clerks, newspaper offices, cigar and news stands, office buildings and at some of the hotels. A card of the same design but somewhat smaller was printed to be hung in 2,000 hotel rooms.

Arrangements were also made with one of the independent

Continued on page 1151

Wing Analysis

Stress Analysis of Commercial Aircraft, Chapter Number Seven

By PROFESSOR ALEXANDER KLEMEN

Donald Guggenheim School of Aeronautics

And GEORGE F. TITTERTON

Chief of the Bureau of Aeronautics, Army Department

IN ANALYZING an airplane the usual order of procedure is wings, landing gear, fuselage (including the engine mount, tail section, and the control system). The order may be slightly varied at some convenient time, but the wing and landing gear must be analyzed before the fuselage as they both impose loads which must be carried through this part of the structure.

Before starting an analysis it is best to collect all the design data that is likely to be required and to list the following items under the heading of General Data:

Aircraft Section		
Gross Weight of Airplane (fully loaded)		
Weight of Wings (to be estimated if not known)		
Net Weight of Plane (Gross weight—Wing weight)		
Total Span of Plane	Upper Wing	Lower Wing
Length of outer section:	Upper Wing	Lower Wing
Length of Bay:	Upper Wing	Lower Wing
Length of Overhang:	Upper Wing	Lower Wing
Chord:	Upper Wing	Lower Wing
Gap (distance between wings)		
Gap/Chord ratio		
Stagger (in degrees)		
Incidence:	Upper Wing	Lower Wing
Dihedral:	Upper Wing	Lower Wing
Area of Wings (including ailerons):		
Upper Wing:	Lower Wing:	Total:
Location of Wing Spans in per cent of chord from leading edge:		
Front Upper:	Rear Upper:	
Front Lower:	Rear Lower:	
Center of Pressure in per cent of chord:		
Upper Wing:	Lower Wing:	
High Incidence:		
Low Incidence:		
Inverted Flight:		
Load Factors required:		
High Incidence:	Low Incidence:	Inverted Flight—
Ratio of Chord to Root component:		
High Incidence:	Low Incidence:	

The required weights must be estimated as closely as possible. This is best done by comparison with similar planes or from knowledge gained in previous designs. At the present stage of airplane development a good first guess of the weight of wings might be as follows:

Wooden Wings		Metal Wings
Lightly loaded airplane	1 lb. per sq. ft.	1.25 lb. per sq. ft.
Heavily loaded airplane	1.25 lb. per sq. ft.	1.60 lb. per sq. ft.
Semi-rigid monoplane	1.60 lb. per sq. ft.	2 lb. per sq. ft.

It is true that metal wings can be designed as light as lighter than wooden wings but only in special cases and after

long experience. For the average American manufacturer the above figures are more nearly true.

All lengths such as span, chord, etc., should be listed in inches for convenience in later calculations. The Gap/Chord ratio is found by dividing the gap in inches by the chord in inches. Gap is the distance between the chord lines of two wings of a plane, measured along a line perpendicular to the chord of the upper wing at any design

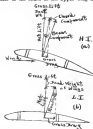


Fig 50

rated point of the leading edge. When the two wings have different chord lengths the arithmetic mean chord is used in figuring the gap/chord ratio. The arithmetic mean chord is the sum of the chords divided by two.

The stagger between two wings of a biplane is the perpendicular distance between the forward third points of the wings. This is illustrated in Fig. 51. In equal chord wings the stagger may be measured between any two similar points, but in planes where the chord lengths are different it is the most convenient to use the forward third point. To find the angle θ , the stagger in degrees, we may connect it off from a side view of the plane drawn to scale. To measure accurately work can measure the distance Y of Fig. 51, divide it by the gap distance at the point and then obtain the θ . θ is the desired angle of stagger in degrees.

Continued on page 1157

Airplane Dopes and Their Application

By GERALD P. YOUNG

"DOPES" is the name that has been given to the cellulose base solutions used as a filling and finishing material for the fabric covered sections of airplanes. A new product which has been developed to meet a need in the aircraft industry which could not be filled by ordinary tanning materials as now when aircraft were being produced, is not new in that of producing a gas film for ballistics and ship logs.

The first gliders were covered only by unfinished fabric and it was due to a desire to make this fabric more airtight the dopes were applied. Starch solutions, glue sizes, and the water soluble finishes were tried. These were greatly affected by rain and moisture to be of any lasting use. Varnish was tried and then, though it gave greater resistance to weathering than the previously used water soluble finishing materials, still did not meet the demands. The dope could not be held sufficiently tight and rigid.

Sometimes used cellulose, a solution of gelatin or cellulose acetate dissolved in a mixture of ether and alcohol. This formed a clear lacelike film, which filled the pores in the skin, but also produced another effect, namely that of carrying the fabric to lightness and produced a smooth, fairly rigid skin conforming to the airfoil section.

The cellulose solution was perfected, using less toxic solvents than ether and alcohol and was the forerunner of the present airplane dopes and the modern airtight finish.

A Study in Itself

The chemistry of preparing airplane dopes or airtight finishes is a study in itself, dependent upon the availability of various organic solvents, gums and plastics. The materials are covered as follows:

Dope is essentially a colloidal solution of cellulose acetate or cellulose acetate, together with cellulose acetate to give a smooth, homogeneous, flexible film when dry, produce tension, and thus increase the strength of the fabric.

A fabric covered surface without the addition of a positive film would be very susceptible to humidity changes, swelling, or tear with every increase or decrease of atmospheric moisture. It would also lose its strength gradually and could not give any protection to the interior of the structure. Coating the fabric-covered surfaces renders them waterproof, tightens the surface to conform to the airfoil section and produces a more nearly rigid airfoil.

The most common compounds of clear dopes are five in number:

1. The cellulose compound.
 2. The gum or lacquer for producing a flexible film.
 3. The solvent for the base material.
 4. The diluent for thinning the solution.
 5. The hardening compounds for controlling expansion and preventing a blooming—that is, provided ventilation is prevented.
- The five compounds are cellulose acetate or cellulose

acetate. Cellulose acetate is the more expensive compound coming to the cost of the raw materials required to prepare it and to its more complicated process of manufacture.

Solvents or diluents are required to produce a flexible film, as the cellulose acetate or nitrate alone would be too brittle.

The solvents are the liquids which are used to dissolve the cellulose base materials. Acetone, methyl acetate, and methyl



A well ventilated dope room. Note that the work is between the operator and circulating fans.

acetate are the most commonly used for cellulose acetate. Acetyl acetate, butyl acetate, and ethyl acetate are the most commonly used for cellulose nitrate.

The diluents or solvents are liquids which have no appreciable solvent power by themselves, but are added to the solvents to keep down the expense of the dope mixture. They separate completely with the volatile solvents. Common diluents for both cellulose acetate and nitrate dopes are benzol, ethyl alcohol, and methyl alcohol.

The high boilers help to prevent whitening or blooming of the dope while it is drying. They tend to hold the cellulose acetate in solution when it is dried under conditions of relative

Continued on page 1156

Last Minute Briefs

The Aero Development Co. of Cedar Rapids, Ia., is developing a new cooled, radial aircraft engine.

The Boeing "Flamingo" biplane flying boat has been tested in the waters near Seattle, Wash.

Major Airplane Products, Los Angeles, Calif., is developing a 120 hp. air-cooled engine.

Arthur C. Chrysler of Duquesne City, Pa., has taken the Travel Air agency for Northern Illinois and Indiana.

The Miller Corporation at New Britain, N. J., is now doing tests for the Travel Air plane for that country.

The first production Halcott engine is being installed in a Fairchild plane in Los Angeles, Calif.

The Champion Spark Plug Co. is producing a new spark plug for aviation engines.

Black Air Services, Inc., plans an extension of its Detroit-Cleveland line to Chicago.

Aircraft Industries, Inc., Oakland, Calif., is constructing three plane hangars to be powered with the Coast 160 hp. engine.

The Washington, D. C., Board of Trade will shortly issue official time tables for air mail plane schedules throughout the country.

Aerovick, Inc., aeronautical engineering company, has announced it has taken new headquarters at Midvale, Ill. The concern was formerly located at Evanston, Ill.

An addition to the factory of the E. M. Laird Airplane Co., Chicago, is to be built. Plant space will be increased by one third and operations will be greatly facilitated.

M. S. Boggs, Aeronautics Branch of the Department of Commerce inspector for New York State, has been assigned to Territory No. 7 and will have headquarters at Detroit.

The British Air Ministry has completed the largest military plane ever built, the "Barracuda" "Indefatigable." It is a monoplane with a span of 136 ft., weighing nearly 35,000 lb.

Two new lighting units for use by airports have been announced by the General Electric Co. One is an airport ceiling light, the other an intermediate field boundary light.

A building program now planned by the Wright Aeronautical Corp. of Paterson, N. J., will increase production capacity of the company to 180 engines a month, according to reports.

Two Air Lines, Inc., a new company in Muskegon, Ind., is now operating a Ford-Stout night passenger plane on night routes near South Chicago with the Ford Airport at Lansing, Ill., as a base.

As AVIATION goes to press, two more airplanes have been granted approval type certification. They are the St. Louis Travel Air biplane model 9006, General engine, and No. 38—Travel Air biplane model 9006, 125 hp. Ryan engine.

Federal patent of an invention of twin tanks to replace the old pistons in airplanes has been granted to William K. Kupper, Milwaukee, Wis. Increased fueling capacity and simplicity, it is claimed, result with use of the tanks.

Three Vought Corsairs have been fitted with amphibious landing gear and are being tested by the Navy. The landing gear is in a single position, with wheels attached to the oleo and a tail skid in the rear.

The Ohio Aero Manufacturing Corp., Youngstown, O., is producing two planes, the Youngstown Youngster and the Aeromaster Corps, two and three plane monoplane airplanes. They will be powered with LeRhain engines.

The Bristol Aeroplane Co., Ltd., Filton, England, manufacturer of the Bristol Jupiter engine is building a 200 hp. six cylinder, air-cooled, radial engine using cylinders adapted to those of the Bristol Jupiter 200 hp., new engine engine.

Umberto Nobile, Italian explorer, has flown his semi-rigid dirigible "Italia" from Milan, Italy, to Berlin-Zeppelin Field, Berlin, Germany, on the first leg of the series of flights which are to take him over the North Pole for scientific exploration.

The Fairchild Aviation Corp. of New York has opened a branch office at 333 E. Michigan Blvd., Chicago, for sales and service purposes. An information bureau is also included. E. W. Fidler, manager, announced.

Charles Levine is building at the Long Island City, N. Y., plant of the Columbia Aircraft Corp., a high wing monoplane to be powered with a Packard 600 hp. engine. The plane is designed by a French and a Russian engineer.

The Short Columbia flying boat, described in AVIATION as work, it is to be used on the Red Sea "A. J. Rogers" air route from Brindisi, Italy, across the Mediterranean to Alexandria, Egypt.

Bert Hall, war-time flyer, has announced he will use a new plane now being completed at Charles A. Levine's plant at Long Island City, N. Y., as a shuttle to Tokyo from Seattle. The plane will be powered with a 37 cylinder Packard engine, it is reported.

The Packard Electric Co., Warren, O., has produced a new variable valve plane to be used in manufacturing aircraft with its various branch offices. The plane was sold to the Thompson Aeronautical Corp., Cleveland, through its distributor for Ohio and part of Pennsylvania.

The Travel Air Manufacturing Co. has completed a new monoplane powered with a Wright Wheland engine. It carries four passengers and 350 lb. of baggage or pilot in 2000 lb. total or approx. Top speed is 125 m.p.h. and landing speed 57 m.p.h.

Philon Aviation, Inc., of Philadelphia has appointed Ed. A. Hollbrook as Director of Public Information of the corporation. Increasing demand of various publications is associated with new lead to the appointment of Hollbrook, who left the New York Herald-Tribune to take up a new job.

That \$1,600,000 of the \$2,000,000 capitalization of the Evans Air Lines has been subscribed is announced by Gen. V. Eakin, president of the new company. The Evans Air Lines has bought out the West Coast Air Transport Co., and will operate a transcontinental line extending from the West and then turning east to Chicago and eventually New York.

'Pirate' Bombers Being Built To be Sent to U. S. Possessions

THE NEW YORK AERONAUT Corp. of Bristol, Penna., is now manufacturing 25 giant "Pirate" bombing planes for the U. S. Government. Fifteen of these are destined to foreign countries, are going to Panama, five to the Philippines, and five to Hawaii.

To ship these bombers, 40 special flat cars will be needed for special cranes for bridges and tunnels. The largest, and will be the largest single government aircraft ship ever made, will consist of 30 specially built cases weighing more than a ton each. Four flat cars will be required in the parts of each plane, and 320,000 ft. of lumber in some cases, 100 miles of three-ply roofing paper, 15,000 sq. yds., 2,000 lb. of steel, 1,000 bolts, 2,500 sq. ft. of heavy wire mesh screening, and 3500 ft. of electric tape will be used in packing. This is in accordance with government regulations, and each crate must be built to weigh as much as it will fit into the hold of the vessel. Each crate must be sealed with two coats of olive drab paint inside and out, a guarantee. The heavy wire screen mesh has been applied to prevent mice from entering the crates and possibly destroying valuable aircraft parts.

Looming Amphibian Approved One Week After Plane is Completed

THE LOOMING Cable Amphibian, which was described in detail in AVIATION for April 6, 1938, was awarded the Department of Commerce Aeronautics Approved Type Certificate No. 34 within one week after the first plane of this type had been completed. Immediately after preliminary sea tests in the waters about New York City and on Long Island Sound, Long Island, N. Y., the plane was flown to Washington, D. C., for the approval of the Department of Commerce. Tests were made under the supervision of Lt. Col. Charles A. Chubb, Jr., Chief of Field, Washington, D. C., and in the presence of the Chief of the Naval Air Station, based with a Port & Whitney Wasp engine, the Cable Amphibian was approved for land or water service carrying a total load of 1100 lb. which includes 1,200 lb. payload in addition to pilot and fuel. As the plane weighs 1750 lb. empty, it has gross weight of 3000 lb. with this load. Immediately after the completion of the Department of Commerce on the amphibian was flown to Detroit, Mich., where it was exhibited at the All-American Aircraft Show.

Inland Eaglerock Co. is Organized In Spokane to Distribute Planes

ORGANIZATION of the Inland Eaglerock Co. of Spokane, Wash., was recently announced. The company will market Packard planes, which are manufactured by the Packard Aircraft Co. of Detroit, in Eastern Washington and Northern Idaho. Besides the territory it will be given.

A term of base of operations will be obtained at once, when a new plane besides the Eaglerock new model, will be delivered. These planes are to be used for aerial fire fighting, training and photographic work.

E. V. J. is president William Hold, vice president, J. W. J. is secretary, and J. W. J. is treasurer. The company is also planning to acquire Eagle Rock and Wilcox General are also planning.

Landing Fields in United States Now Reported to Number 5,000

LANDING FIELDS in the United States now number 5,000 according to a statement issued by the Aeronautics Branch of the Department of Commerce. A list of 1,075 airports has been made public by the department, and with nearly 4,000 others available for landing, the country now offers numerous facilities for the ever increasing number of pilots.

Many of the fields offer landing and take off areas alone, but 231, on the other hand, are fully equipped with the latest aeronautical aids for day and night flying. California, with 115 air fields, leads all other states. Eight of the 115 are marked and lighted by the Department of Commerce along regular airways and some others are also equipped for night flying. Texas is second with 90 fields, only two of which, however, are maintained by the Department for night flying, and these others similarly equipped.

Pennsylvania, third in rank with 69 fields, has 24 airports fully equipped and maintained along airways by the Department, more than any other state. In addition Pennsylvania has two other airports for night flying. Illinois with 50, of which 22 offer facilities for night landing, and Ohio and Oklahoma each with 45 fields, follow. Wyoming has the highest proportion of most modernly-equipped fields, 17 of 21 fields offering all night work.

Intermediate landing fields along regular air mail routes in the United States, marked and lighted by the Department of Commerce, are numbered at 123. In addition there are listed 38 commercial or other fields, equipped with rotating beacons and partial or full equipment of flood lights for landing, flood-lighted buildings, secondary lights, and danger lights.

Port Angeles, Wash., Business Men Will Pick and Build Own Runway

AN AIRPORT built liberally by civic enterprise can be located by Port Angeles, Wash. The town treasury was low, but it proved an obstacle in the path of a determined group of business men and civic leaders who wanted to waste no time in placing the city on the aviation map.

Dancing events, among themselves with pilots, shovels, and other implements, the Port Angeles business men and civic leaders are organizing the Rotary, Kiwanis, Young Men's Business Club, and Chamber of Commerce, recently used to work building a runway on the airport site. Before nightfall the 75 workers had completed the leveling and grading of a runway about 500 ft. long and wide to permit air planes to land or take off. Mayor W. B. Dinkich and Port Engineer W. J. Murphy were in charge of the work.

Los Angeles-San Antonio Passenger Line to be in Operation by June 1

REGULAR DAILY passenger service between Los Angeles and San Antonio, Texas, will start not later than June 1, according to a recent announcement by W. C. Eberly of the Pacific and Gulf Airline, a Los Angeles concern organized to install this service.

Eberly said that a trip will be made once every day on a schedule of 22 to be stopped there. It was said that the company will operate 10 to 15 passenger biplanes and two air passenger single engine biplanes has already been ordered.

Production Started by A. V. Roe On Avian III Using Cirrus Engine

A. V. ROE & Co., Ltd., Newton Heath, Manchester, England, recently announced that it was starting production on a new model of the Avian "Avian," as it is known as the Avian III. The new model has a smaller wing area, namely 244 sq. ft. and is of smaller construction than the Avian II being somewhat lighter. The Avian III weighs 812 lb. empty compared with the Avian II which weighed 826 lb. empty. The new model, because of its reduced wing area, carries a lighter load and weighs 1380 lb. loaded, while the Avian II weighed 1408 lb. loaded. Both of these planes are powered with the Cirrus II engine rated at 75 hp. at 1,800 r.p.m. and developing 80 hp. at 2,000 r.p.m. With this engine the Avian III is stated to have a high speed of 200 m.p.h. at sea level and a high speed of 160 m.p.h. at 5,000 ft. The plane has a cruising speed of 60 m.p.h. and is said to have a run of 80 to 200 yd. after landing. It takes off at about 70 ft. The climb at sea level is 600 f.p.m. and absolute ceiling is 17,000 ft.

The Avian III is of wood construction with steel struts and fittings. The highest wings have rounded I section upper ribs with rounded upper and square cap struts. The wings are mounted on the gliders by being dished, reducing the overall width of the plane to 9 ft. 2 in. The fuselage has wood longerons with the sides covered with plywood, except at the wing housing. The cockpit is in tandem with the controls movable as a unit. They are all mounted on a removable panel that sits on the floor. The Avian has a span of 28 ft. and a length of 24 ft. 10 in.

Pioneer Instrument Co. Receives Two Large Orders and Contract

RECEPTION OF large orders for instruments from the Fairchild Aviation Corp. of New York and from the Mohony Aircraft Corp. of San Diego, Calif., has been announced by the Pioneer Instrument Co. of Brooklyn, N. Y. A contract, furthermore, has been awarded to the Pioneer company to supply a complete line of new instrument instruments for installation on planes manufactured by the Bellanca Aircraft Corp. of Wilmington, Del.

The order from the Fairchild company calls for 150 sets of various instrument instruments consisting of altimeters, barometric level indicators, compasses, and air speed indicators, while that from the Mohony company specifies 150 sets of barometers, altimeters, magnetos compasses, and barometric indicators.

Airplane Dopes and Their Application

Continued from page 1149

travels high humidity and in the presence of the water that forms on the surface owing to the shifting caused by expansion of the highly volatile solvents.

Besides producing, when dry, a taint of the fabric that is not greatly affected by changes in atmospheric moisture, dopes should act as a protective coating to prevent rapid deterioration from weathering. A slow film of relative adhesion or cohesive moisture dope is transparent to sunlight, and sunlight is the greatest factor in the deterioration of dopes and fabrics.

To prevent excessive deterioration, an opaque, non-oxidizing coat on top of the clear dope film is of the utmost importance. This opaque protective coating is a dope made from the clear dopes already mentioned, but with the addition of an inert opaque pigment that cuts out the action of sunlight from the clear dopes and fabrics. A clear dope is only subjected to a few months' constant exposure to the sun, while the same fabric protected by an opaque pigment dope will last the several years.

There has been prepared within the past few years a few systems of doping preparations that will give equal protection, produce equal or better finishes, and produce a simple method of application than that mentioned above. The opaque solvents consist in applying four coats of a very fine dope to the fabric and finishing them with a glass dope of special fabric longer.

The use of this solvents terminates the use of other solvents or clear solvents like dopes. The pigment content of the non-pigmented dope is sufficient to produce adequate protection from surface light with four coats, and at the same time an excellent pigment material to affect the finishing, dyes, dyes, dyes, or taints of the dopes.

To produce a finish of maximum durability and maximum weight four coats of the non-pigmented dope will give a finished dope a weight of 1.5 to 2.0 times that of the clear dope. This is less than 2.5 percent per square yard weight of dope and may be as low as 20 percent per square yard.

Four Coat Scheme Less Costly

This four coat scheme will be less costly than a previous one-point, as it will require less dope material and also the cost of labor to application four coats costs less than one coat. Only when the question of maintenance of finished dope and a high glass finish is desired is it necessary to add another. This of course being an obvious condition, for a product, the highly smooth glass finish is necessary to apply sufficient film to completely fill all inter thread spaces in a uniform film produce a satisfactory finish surface overhang. Should be considered.

The fabric should be stretched to a uniform degree of tension throughout the surface. This tension can be determined by by hand as the dope will add to that already over the fabric and be applied and is loaded only by the strength of the structure, in other words, it is not necessary to stretch the fabric to the tension of the application, but it is stretched too tightly at the start the dope will cause a moderate distortion of the wing structure where it has thoroughly dried.

To produce a surface that will have the best results of uniformity and maximum change in atmospheric humidity the surface should be kept dry, if possible, the surface is covered with fabric. If the fabric is not held tightly over the structure and then not been for a short time, it assumes a permanent set and does not return to its original position. This is not a satisfactory finish. Changes in atmospheric humidity will also cause the fabric to warp and if it is allowed to warp several days before doping.

The first two coats of dope should be applied by brush and subsequent coats by spray gun for best appearance and speed of production. This is true for both clear and non-pigmented dopes. The first coat should thoroughly wet the fabric, but should not be allowed to dry. The second coat should be applied with brush or spray gun before drying. It has been found that the first coat of dope should be applied with brush or spray gun and the second coat should be applied with brush or spray gun.

Naturally four coats of dope will produce a surface of uniformity and a high glass finish. When it is desired to

a product a high glass finish, which adds only to the value of the plane in appearance, care should be taken in securing only well colored finishing longerons are an especially good to insure flexible when applied over fabric.

The present dry atmospheric exposure, while entirely suitable for the over any metal surface, are too liable to be of any value when applied to dope fabric.

There are several ways of producing a glass finish. The most method has been to varnish over the dope surface with a special glass over varnish. This produces a very high gloss, and is not entirely satisfactory. The best grade of varnish will fill under weather exposure and shrink away rapidly from the pigmented glass longerons. It is also liable to make a good patch over a varnish surface.

A second method is to mix with the unpigmented dope and for materials a glass liquid which when sprayed on a finished wet produce a glass surface. An advantage of this type coat is that it is liable to give a more uniform finish than any other type of glass surface may be applied with a brush or spray gun. This method has been used for the finishing part.

A third method of finishing is to apply over the clear dope a non-pigmented dope undercoat a finish of lacquer finish is somewhat longer, but made more flexible by the addition of a small amount of solvent. There are several of the most progressive dope and lacquer manufacturers who are at present producing this type of aircraft finishing material.

Cost Figured from Area to be Finished

The ends of various dope schemes can be figured from the use of the surface to be finished. Clear and non-pigmented dopes will cover about 110 sq. ft. of area, one coat, per gallon, when applied by brush, and when properly thinned for spray application unpigmented dope, pigmented dope and lacquer will cover about 170 sq. ft. of area, one coat, per gallon.

All doping of airplane surfaces and spraying of metal parts will depend should be done in a well ventilated room. The temperature should be kept at about 70 deg. or cold weather.

All brushes, when not in use, should be washed out with kerosene or gasoline. Spray equipment will not give satisfactory performance unless cleaned thoroughly when each day's work is finished. Special care should be taken to clean the spray gun, tank, hose, and nozzle, which should be washed out with kerosene.

Any good lacquer spray gun will handle dope materials but no experience is necessary before speed and uniformity of finish can be expected.

A good painter usually makes a poor dope man for brush application, for where paint and varnish should be well worked in when applied, dope should be worked as little as possible and kept left as much as it will hold without dripping from the brush.

When necessary to make a patch over a dope surface certain precautions should be taken. Wash the area with dope solvent, if necessary, to remove any old grease or varnish which may be on the fabric. This thinner will soften the dope film. The thinned surface coats should be worked off, with a brush, and then a patch of the same dope should be applied in a layer that will be about 1/8 in. thick. The patch should be allowed to dry in a warm place for at least one and one half inches before the work is continued.

The edges of the patch should be worked or sanded about 1/8 in. or more all around. Dope should be applied to the surface of the wing as well as over the patch. The edges of the patch should be worked off, with a brush, and then a patch of the same dope should be applied in a layer that will be about 1/8 in. thick. The patch should be allowed to dry in a warm place for at least one and one half inches before the work is continued.

Wing Analysis

Continued from page 1148

The location of the wing apex as per cent of chord from the leading edge must be figured. If we divide the distance in inches of the span from the leading edge by the chord of the wing in inches and multiply the number by 100 that will give the desired distance as per cent of chord.

The center of pressure of many commonly used wings is located at the various lift conditions in Table 25. For wing sections not listed the high incidence and inverted light order of pressure must be taken as its most forward position; the low incidence position is taken as the position at the angle which is 0.5044. The center of pressure will be readily determined from the wing tinned data on airfoil.

TABLE 25

Center of Pressure Location and Ratio of Chord to Mean Chord for Various Airfoils
C. P. Location in % of Chord

Airfoil	Mean Chord	Ratio of Chord to Mean Chord	Center of Pressure
Acromechanic 3A	10.1	1.0	25.0
Acromechanic 3B	10.1	1.0	25.0
Acromechanic 3C	10.1	1.0	25.0
Acromechanic 3D	10.1	1.0	25.0
Acromechanic 3E	10.1	1.0	25.0
Acromechanic 3F	10.1	1.0	25.0
Acromechanic 3G	10.1	1.0	25.0
Acromechanic 3H	10.1	1.0	25.0
Acromechanic 3I	10.1	1.0	25.0
Acromechanic 3J	10.1	1.0	25.0
Acromechanic 3K	10.1	1.0	25.0
Acromechanic 3L	10.1	1.0	25.0
Acromechanic 3M	10.1	1.0	25.0
Acromechanic 3N	10.1	1.0	25.0
Acromechanic 3O	10.1	1.0	25.0
Acromechanic 3P	10.1	1.0	25.0
Acromechanic 3Q	10.1	1.0	25.0
Acromechanic 3R	10.1	1.0	25.0
Acromechanic 3S	10.1	1.0	25.0
Acromechanic 3T	10.1	1.0	25.0
Acromechanic 3U	10.1	1.0	25.0
Acromechanic 3V	10.1	1.0	25.0
Acromechanic 3W	10.1	1.0	25.0
Acromechanic 3X	10.1	1.0	25.0
Acromechanic 3Y	10.1	1.0	25.0
Acromechanic 3Z	10.1	1.0	25.0

The ratio of not chord (percentage) in not been computed for all airfoils as 0.5044 is not included and not included in the table.

The center of pressure location for "Stable" airfoils in the low incidence condition should be obtained from the table. The center of pressure for airfoils in the high incidence condition should be obtained from the table. If such data are not available, a location between 45 and 60 per cent of the chord should be assumed. It has been found that a downward action of airfoil as flap moves the center of pressure far back and when a "Stable" airfoil is designed on the forward location determined by wind tunnel tests without flaps, or flap action, the rear edge of the wing is usually very flexible.

The load factors required must be determined from Fig. 1 of Chapter 3. The high incidence factor is determined from the curves and low incidence is taken as 65 per cent and inverted light as 90 per cent of the high incidence value.

Before finding the ratio of chord to mean chord in the General Data it is best to understand why there is such a ratio. The lift force on an airplane acts at right angles to the wind striking the wing and the drag force acts parallel to the wind. This is illustrated for high and low incidence flight in Figs. 30a and b respectively. In low incidence the lift force is nearly coincident with the beam force that they are taken as equal. This is, the net beam force equals the gross lift force less the drag weight of the wing. In high incidence the weight of the wings imposes any additional stress upon themselves and so their drag weight is subtracted to obtain the force that will be used in all later calculations, namely, the net beam force.

In high incidence the lift load and the beam load are at greatly different angles. As in low incidence the drag weight of the wings is subtracted from the gross lift load to obtain the net lift load. The net lift load is based on the total or gross drag load and the net lift force. This force is

however, is not perpendicular to the beam and therefore is not equal to the beam load. It may however be broken up into components, one perpendicular to the beam (not beam load) and the other parallel to the chord (not chord load). It is to be noted from Fig. 56b that the drag component or net chord load acts toward the leading edge of the wing. In the low incidence condition, Fig. 56a, the drag or chord component acts back. Thus in high incidence the chord component is considered negative and in low incidence positive.

The ratio of net chord component to net beam component is always two in inverted flight (in which condition it is assumed there are no chord loads). In low incidence the ratio is always taken as 0.15 as this is fairly close to the true value for all airfoils. In high incidence the ratio of chord component to beam component may be obtained from Table 15 for cambered and airfoils. For sections not listed in the table the ratio may be obtained from the formula:

$$C/B = \tan(\alpha - \beta)/2 = 0.15/D$$

$$\text{where } C/B \text{ is the ratio of chord component to beam component}$$

α is the angle of attack when the center of pressure is at its most forward location

β is the angle of the wing at the angle of attack, l , found just above

Example: Clark Y—To determine the ratio in high incidence

$$C/B = \tan(\alpha - \beta)/2 = 0.15/D$$

$$\alpha = 14 \text{ degrees for most forward location of C.P.}$$

$$\beta/D \text{ at angle } \alpha = 14^\circ = 9.9$$

From tables of trigonometric functions: the angle whose tangent is 9.9 is 50 degrees.

$$C/B = \tan(50^\circ - 14^\circ)/2 = 0.154 \text{ which is the value listed in Table 15 for the Clark Y airfoil.}$$

The Department of Commerce, the Army and the Navy all advise that the wing analysis should be made in the following manner:

- 1.—Compute or estimate the gross weight of the plane.
- 2.—Compute the effective span of each wing.
- 3.—Compute or estimate the dead weight of the wings.
- 4.—Compute the relative efficiency of the wings. (See Fig. 55.)

- 5.—Compute the normal gross beam load on each wing.
- 6.—Compute the normal net beam load on each wing.
- 7.—Compute the normal chord load on each wing.

When this has been done the loads on the wing are distributed between the spans and the moments and reactions fixed. The stress in each span, steel and wire of the wing outline is wired for and a member strong enough designed to take such stress.

The computations of the above for the conventional type of design may be made as follows:

- 1.—The gross weight is taken from the General Data published previously.

- 2.—The effective span of a wing of constant chord and section is the distance from wing tip to wing tip, less that portion of the length, of wing, that is covered by the fuselage and nacelles, and an allowance for tip loss. The effective span span S_e in Figs. 51 and 52, is one-half of the effective span L_e . The entire span of the airplane is not effective because at the tips the air forces are greatly reduced. This is caused by the air slipping off the tips. The wing then does not receive the full force of the air in this region. The method of determining the decrease in effective span to allow for tip loss is shown in Fig. 51 for externally braced and in Fig. 52 for internally braced wings.

For externally braced wings a length equal to $1/4$ of the winging (10.24 ft.) is subtracted from the span to obtain the effective semi-span; for internally braced wings a length equal to $1/10$ of the tip length (10.24 ft.) is subtracted. In the latter case L_e , the tip length, is equal to the chord for rectan-

gular wings and for tapered wings it equals $1/2$ chord at its length just equal to the chord at that section.

The length of span occupied by the fuselage is $1/4$ chord is subtracted directly from the actual span to obtain the effective span.

- 3.—The dead weight of the wings is taken from the General Data.

- 4.—The relative efficiency of the wings is determined from Fig. 55. This figure gives the ratio of the normal lift on the upper wing to that on the lower wing as a function of the gap/chord ratio and stagger in degrees. When a wing of a biplane is close together the upper wing interferes with the lower wing destroying some of its lift. The upper wing is more effective for low gap/chord ratios. Staggering the wings positively (making the lower wing the rear) makes the upper wing even more effective. It is

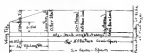


Fig. 55

preferable from a structural viewpoint to have both wings equally effective. It is not important enough a consideration however to interfere with vision or accessibility to the cockpit.

When two wings of different airfoil sections are used the chart may be used of the wings here about the same characteristics. With other combinations a wind tunnel test must be made as a very complicated theoretical study.

In using Fig. 55, the gap/chord ratio and the stagger as measured on a vertical section midway between the side of the fuselage and the effective tip of the shorter wing. The same relative efficiency shall be used for all flying conditions.

- 5.—The normal gross loads for each wing are to be obtained from the formula:

$$W_u = \frac{W_g}{2} + L_u \quad \text{and} \quad W_l = W_g - W_u$$

Where W_g is the gross weight of the airplane in pounds
 e is the efficiency of the upper wing as obtained from Fig. 55

L_u is the effective span of the upper wing in feet
 L_l is the effective span of the lower wing in feet
 W_u is the normal gross beam load on the upper wing in lb. per sq. ft.

W_l is the normal gross beam load on upper wing in lb. per sq. ft.

W_g may be taken as half the gross weight and L_u and L_l the effective semi-spans in ft. and sq. ft. desired.

- 6.—The normal net beam loads, W_u and W_l , are found by subtracting the dead weight of the wings W_g as given from W_u and W_l respectively. The dead weight per sq. ft. of wing is found by dividing the total wing weight (the internal struts and wires) by the total span of a wing in inches before any decrease due to tip loss is made. In other words the actual span is used and not the effective span.

Thus $W_u = W_u - W_g$ and $W_l = W_l - W_g$

In designs where the chords of the two wings are (probably) different a dead weight per each wing must be found separately. This is seldom the case however.

When the normal net beam loads have been found it is possible to check the work up to this point. Thus a dead W



Fairchild Cabin Plane (Wasp)
as used by the Canadian Transcontinental Airways

Reserve Power

RESERVE POWER in all aircraft is of great importance, and a necessity in commercial aviation. It provides a greater factor of safety, increased dependability, and longer life. An important step in American commercial aviation of 1928 will be recognition of the factor of RESERVE POWER.

"WASP" engines in the Fairchild monoplanes of the Canadian Transcontinental Airways provide the necessary RESERVE POWER. Quick take-off, with the combination skis and pontoons with which this ship is equipped, is vitally essential in the transportation of the Air Mail from the mouth of the St. Lawrence River to Quebec.



THE
PRATT & WHITNEY AIRCRAFT CO.
HARTFORD CONNECTICUT

DEPENDABLE ENGINES



For The Discriminating
Purchaser:

THE AIRSEDAN

AIRLINE OPERATORS will find this plane meets with all their requirements. The cabin has exceptionally comfortable seats for four large passengers and the pilot's visibility is unexcelled.

PRIVATE OWNERS will approve of the fine appointments which are selected to satisfy the most critical taste.

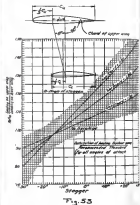
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tip loading to be formulated. Then an arbitrary wing will be necessary.

For externally braced monoplane or biplane wings, the center wing panel is at a distance materially greater than chord length from the tip, the following method of wing must be used. As before for wings where b is the span, the load curve of Fig. 51 must be used. For stresses outward of this section point, extend b by increasing the loads obtained from Fig. 50 by 20 and 10 per cent, the loading of Fig. 52 must be used. This is a wing where loading and gross stresses in terms of Fig. 51 are outward of the section point. The tip length in the load



ing of Fig. 52 extends one chord length in. In the inverted wing it ends at the section whose chord just spans the distance from the tip. From this point the loading from its full value to 0.5 of this value at the tip. The loading of Fig. 52 must be used for moments and load from the spar from its outer point of attachment to the tip for the loads on the outer struts and flying wires.

For externally braced wings (the full resultant moment load curve represented in Fig. 50 should be used. The loading is the most severe for this type of plane.

When the proper loading has been decided for the proper under consideration it is best to assume a load of 1 lb. per sq. ft. and to solve for moments and stresses. On the other side, such a load. When the actual span load is used for each flight condition the moments and stresses are obtained for a unit load may then be multiplied by the actual load for each condition. This gives a set of work as the tip load is solved once. Chapter 2 of this book is an excellent way to solve spans for moments, shears and react. etc.

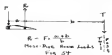
As the normal net beam load as a wing has been found as a unit, the unit load is used between the spars. With proper construction the beam load on the wing is assumed to act at the center of pressure and must be divided between the spars in inverse proportion to their distances from the center of pressure. From Table 15 we have already found the center of pressure. The center of pressure location for the various light conditions. Likewise the spar locations are listed in per cent of chord from the leading edge. Then for any condition:

For upper spar location — center of pressure
For lower spar location — front spar location
For upper spar location — front spar location
For lower spar location — front spar location
For upper spar location — front spar location
For lower spar location — front spar location

The normal beam loads on the wing are multiplied by the above percentages to obtain the proportions of loads on the front and rear spar. These percentages will of course vary for each flight condition as the center of pressure varies. For high incidence and inverted flight the center of pressure is assumed the same but the inverted flight loads act down and the high incidence loads up. Thus far no load factor has been considered. The loads on the spars in each condition may now be multiplied by the load factor for that condition. These factors have been listed in the General Data and are very handy. The loads which we now have for each spar for each flight condition is the actual beam load. If we have selected our spar for a unit loading we may now multiply the results by these actual beam loads. In this manner we have the moments and reactions along each of the spars for all flight conditions. If our plane is such that the outward stresses and moments are to be increased 15 and 50 per cent.

we may now do that and then obtain the design load in spar at the strut point and the design load in the outer struts and lift wires.

Application of Chord Loads to Internal Wing Truss
The chord loads are considered as concentrated loads at the panel points of the internal wing truss or drag truss as it is more commonly called. The load at each wing point is



the product of the normal chord load per inch run at the point and the distance between midpoints of adjacent panels of the internal truss. For this purpose, the extreme wing tip shall be considered the midpoint of a truss panel.

Thus far we have discussed the method of obtaining the loads due to high incidence, low incidence, and inverted flight. The condition in which the plane is nose-diving must be investigated also. As discussed in Chapter 5 it is quite difficult to know just what is happening in a dive. For purposes of stress analysis however the following condition must be used.

In the nose-diving condition the beam loads on the front spar must be the same as the design ultimate loads on those spars in the inverted flight condition. In other words after the

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load factor and whatever other restrictions were applied have been included in the inverted flight loads on the design. These loads are taken on the design loads on the Y. 1. No further load factor is introduced. Then to obtain the loads on the rear spar moments are taken about the Y. 4. This is illustrated in Fig. 54. For equilibrium these loads are acting upward. But a load is set down on the tail spar to the difference in the spar loads. This momentary load and counterweight Y is obtained and as there is no force in a horizontal plane the whole system balances.

The upward load on the rear spar is distributed to the rear section. That is to say we may multiply the moment and moment obtained by the rear load on the spar by the new beam load on the rear spar and obtain the actual moments and reactions.

The chord components of the air load on the wings is equal to the net weight of the airplane. The load per unit area may be determined by dividing the net weight of the plane by the actual area of the wings. The net weight of the plane is the gross weight minus the wing weight.

For internally braced cantilever monoplane the loads applied there must be increased 50 per cent.

Wings dead weights are caused by the wings such as leading edges, wingtip floats or struts, or other items that weigh little and are in such locations that they do not affect the stresses much, their weight may be included in the dead weight of the wing. Loads that are large or are so placed that they would make an appreciable change in the moment stresses such as gas tanks, engine assembly, or wing floats for outboard of a strut point must be separately provided for. If the load is concentrated at one point it must be proportioned between the span at that point in terms proportion to the distance of its center of gravity from its span. These spar loads must then be treated as concentrated loads and the effect on the stresses calculated. If the net is spread over a considerable extent of wing area, like a gas tank or a wing radiator, it shall be considered as composed of concentrated loads at its points of attachment to the span, as if a uniformly distributed load if it is attached to the span at sufficient points. If a dead weight is supported in such a manner that it may be considered as uniformly distributed and extends over two-thirds or more of a wing panel it may be considered as distributed over the entire panel. This makes the computations considerably easier. When the net is one of variable weight such as a gas tank the primary analysis must be made with the maximum load and then a rough check must be made with the light weight to see that nothing is designed by this condition.

The weights of all dead loads shall be assumed to act in the beam direction to the blade and low loadings and careful flight conditions, and in the chord direction in the rear direction.

Summary.

This chapter summarizes very nearly all the rules of wing analysis. In the next chapter a sample analysis will be worked out to illustrate the operation of these rules.

Copyright Alexander Klemin

To be continued on the next issue of AVIATION

Potomac Flying Service

Continued from page 1147

See companion for the Potomac Flying Service to see the best way to the airport of any one who will take on an plane ride. Another interesting time is spent in the field are on the side of the company. Whenever a pilot driver receives a letter to "see" Washington, they usually follow an indirect suggestion for a flight over Wash-

ington. "Specially when it can't give you a thing to get out there." The companion works quite alone.

It is a well-known fact that tourists are more numerous for each of the interesting places they visit if for an other reason than to send them to their friends back home. Potomac has been pointed, one showing a fine place on the plane and under with the picture of Hoover Field. Nearly every person taking a flight says he or more of those cards. First it is a source of interest and then it is one of the best forms of advertising throughout the country.

Personal Contacts. Two men were employed to see persons only—to create a personal relationship between the man and woman who come in contact with prospective flight customers and the Potomac Flying Service. These men are known to the public. Their business is to make good will towards the customer. They visit the hotels, talk to the clerks, direct the tourist agency, arrange for free rides for all those who are taking the flight tickets, volunteer to help in the sale of the tickets, show themselves ready to give their time to make the sale in a particular place where the reserve from the sale of tickets is large.

They meet the buses, distribute the tickets, talk to the business, that with the taxi drivers and all the time they talk "By Over Washington". That's the slogan of the service. Whatever they do and say, it is always to get more people to fly over Washington. The personal interest these men show in the ticket sales made by various places and persons has a beneficial effect. It is a good idea to let others that the Potomac Flying Service is not only interested in distributing these tickets but in helping them to make money for themselves. When the leaflet and other advertising matter is sent the service men complete by their paid and helped personally.

Freezer Publicity. "We can get enough business without any more publicity." This does not mean that we do not

believe in newspaper publicity," stated Mr. Mack. "I believe that the newspaper publicity we received from the Washington papers brought out many residents to take a flight and also brought us many orders."

"We are careful, however, what kind of publicity goes into the papers. I have no doubts that these 'stories' about saving stars, birds and drops only interested the belief of the average man, that for flying one must have a special mental and physical equipment. What we are trying to get across is that any one with good eyes and common sense can fly. Of course some think that even for a night-long flight one has to be a special type of person. And I do believe that some of the 'stunt' publicity is responsible for that belief."

Give Reporters Free Rides

"The newspaper reporters are always given the latest information regarding new types of planes, new inventions, and prominent people who are taking up flying. Very often we give them free rides. Recently one of the reporters was given a ride to New York. This may be a better idea about flying and therefore become a better reporter about flying."

Mr. Burdett realized that the favorable impression created by the various forms of advertising and publicity must be maintained and even strengthened after the prospective passengers or people arrive at the airport.

The company erected a small but attractive combination office and waiting room of light bryon lark with white trim. This pleasant looking building is the first thing that strikes the visitor. On entering the building the visitor finds a neat waiting room partitioned off from the office by a railing. Clean chairs, a table with magazines, and a spring water fountain are there for the visitors. Not a trace can be found of broken glass, spill, propeller, oil containers, swabs of oil, waste or greasy flying logs frequently found in the

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Side Slips

By ROBERT S. OSBORN

Mr. N. V. G. tells a story, worth passing along, about a pilot who was recently assigned to ferry a very old plane on a rather long cross-country trip. He had worried his engine up, moved the cockpit away from the wheels and was about to start out when a final inspection of his instrument board showed that there was no clock mounted there. "Say," he said, "There's no clock in this ship and I haven't my watch with me. How am I going to tell time?" None of the standing about the plane had any response to offer, so finally he said to a mechanic: "Well, I've got going off without any way of keeping track of time. Run on in the office and get me a calendar."

Mr. C. W. H. of St. Louis, happened about a month later in Aviation concerning the new Ford airplane engine, in which was the statement, "the loss of this engine is 400,000 when throttled down to 1400 rpm." The only possible explanation that seems to be offhand, C. W. H., is that the engine becomes more bored when it is held down to its low speed.

Because of the worries of the newspapers and various governmental bodies, the King of Belgium has given up on an airplane trip he expected to make to a foreign capital. Reminded of an old saying,—"Daddy has the best of the worst a crowd,"

Cozy Jones turned up late at a party recently with lady Motters and got hands and a good story to explain her being there in that condition. He had made a trip over to Yer Jersey State New York on a rather long trip. No one was to come back with him on the return trip and not having a engine starter on the plane, he isolated the aid of a nearby farmer for help in getting started. The farmer was placed in the cockpit to operate the switch while Cozy came to the propeller. For forty minutes he swung the prop, and propeller and cleared the engine, tried different combinations of levers and different ways of swinging the propeller, without even a thought out of the engine. At the end of that time he was about to give it up altogether when he discovered that the farmer was back having turned the gas cock off at the end of the time.

A New York manufacturer of women's dresses is looking that he recently shipped two hundred gowns to Toronto by airplane. To our way of thinking a load of two hundred gowns is hardly worth mentioning, either from the angle of weight-carrying for the airplane or as a man job. If the manufacturer had any thought to have sent a woman's address on the trip with the announcement that she had left her yard work alone, he might have gotten a lot more publicity out of it.

The Editorial Assistant was in the office the other day in connection with the production of those famous "Motters" president of The Canadian Aviation League, that a letter for distant future was would have airplanes with a weight of 1000 lbs. He says that this is actually possible by operation, as he has a plane now which makes 1000-lb. loaded miles per hour, and he has a plane which makes 1000-lb. in the other direction, and if he could make a single trip a day, this production would be reached right away.

FOREIGN NEWS

*In Special Arrangement with the Transportation Division
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A Japanese Flying School

Under authorization of the Japanese Government, the Nippon Aviation School has been organized and opened in Tokyo as now being given at the school's support at Tachikawa on Tokyo, reports Tetsuo Aiba, president of the school. The instructors are K. Ogawa and J. Yoshida, both of whom are experienced pilots.

French biplane and British Avro planes are used by the school as well as a Zivko type plane designed and constructed



Students on the airport of the Nippon Aviation School near Tokyo. Biplane and Avro planes are used up mostly for flight, while a Zivko type plane designed by the Japanese is shown in the foreground.

is the Japanese, which is powered with a Hispano-Suiza 200 hp. engine. A move of students is enrolled for the month, six of them being a young Japanese woman of 20.

French Air Clubs Hold Sixth Congress

French air clubs recently held their sixth national congress at Lyons, it is reported. Conference held were chiefly aimed to the discussion of the general conditions affecting French commercial aviation, resolutions being passed without loss of the policy of the organization of clubs toward this end.

A request for governmental assistance in keeping about better meeting facilities for pilots was included in the most item passed. Other measures favored a program for the job of conditions for furnishing the air and aviation, an development that more and better airports are needed in view of the personal rates, and a request for an air service from London to Marseilles that would permit mail to fly the fastest city by plane at 2 P. M. and arrive in Marseilles at 6 A. M. the following morning.

German Makes Record Ascent in Glider

In about 1,600 ft. above his starting point is 45 m. per hour and he held a record of 100 m. per hour at 100 m. in Germany. The glider climbed north, which was made in 1937, last October, at 2,000 ft., but this height was only reached after more than 4 hr. in the air, therefore the mark 1,600 ft. to 45 m. is a new record for speed of ascent to 45 m. height.

Vietnam Now Has Aviation Department

A decree creating a Ministry Government Department of Aviation was recently approved by President Calles. The decree is to have jurisdiction over the development and the maintenance of all airports, and the regulation of all things which are aviation training.

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